

# **Viscoplastic Models and Finite Element Schemes for the Hot Rolling Metal Process**

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# Viscoplastic models and finite element schemes for the hot rolling metal process

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**Abstract.** In this work, pure viscoplastic models and their finite element discretization employed for the hot rolling method of plate metal forming are discussed. The derivation of the viscoplastic flow rule and the relation between the stress-strain rates are presented. The associated equilibrium system of the deformation problem is completed by describing the form of the inlet/outlet boundary conditions as well as by introducing the contact conditions and a viscoplastic friction law. The system is discretized using the continuous finite element method where classical penalty terms are used for incorporating the boundary conditions and the contact conditions. The use of a viscoplastic friction law requires an estimation of the friction parameter which is obtained through the derivation of stability estimates for the finite element scheme. Numerical tests are performed and their results compared in order to investigate the effect of the strain sensitivity parameter on the behavior of the plate velocity and on the magnitude of the interfacial stresses

**Key words:** hot rolling, metal forming, pure viscoplastic models, Norton-Hoff laws, contact-friction interface conditions, finite element discretizations, friction parameter, boundary constraints, penalty terms

## 1 Introduction

In recent years, there has been a growing interest in the use of accurate mathematical models and numerical methods for describing and simulating the rolling metal forming process, i.e., the process of the plastic deformation of a metal plate passing through a pair of rotating rolls, [10], [13], [16], [17], [18]. During the rolling process, the thickness (height) of the metal plate is decreased while its length is increased. The rolling technique for metal forming involves many phenomena such as nonlinear plastic deformation behavior, large strain rates and depends mainly on the friction and contact boundary conditions. The development of efficient and accurate finite element methods for studying these problems is an interesting subject that has attracted many scientists.

In the work [37], viscoplastic models were used and the frictional stresses were computed using a thin layer of elements across the interface. An investigation using rigid-plastic finite element methods and employing different friction laws in the rolling process of metal forming was carried [19], and in a later work [12], rigid-plastic models were used including the variation of the rolling coefficient across the interface. In the study presented in [25], viscoplastic strain-stress relations were applied by assuming two different friction models. A series of numerical tests were performed for computing the tangential and normal stress on the interface points. In recent years, other kind of models and many other different finite element methodologies have been proposed. Mesh free finite element methods for slightly compressible rigid-plastic models have been applied for simulating plane strain rolling problems in [34]. Hybrid Eulerian-Lagrangian finite element formulations for discretizing rigid viscoplastic models have been discussed in [27] and [29]. In [27], a numerical investigation of the rolling problem was presented by performing several tests using different values for the radius and the velocity of the rollers. In the recent work [30], elastoplastic-models and Augmented Lagrangian techniques have been used for discretizing the contact and the frictional conditions. In addition, the authors proposed an optimal control problem for achieving the desired shape of the plate. Rigid-plastic finite element schemes with efficient and fast nonlinear solvers were developed and analysed in [35].

47 The motion of the plate is mainly governed by two equations, the constitutive equation and  
 48 the continuum equilibrium equation. The constitutive equation provides a relationship between  
 49 the stress and strain rate (deformation rate) [24], [2], [31]. In many metal forming processes, the  
 50 elastic effects are weak and viscoplastic constitutive relations are introduced for describing the  
 51 problem, which lead to an Eulerian descriptions with the velocity components as the unknowns.  
 52 Here, we consider a pure viscoplastic (Norton-Hoff) model, such that the constitutive equation  
 53 has the form  $\boldsymbol{\sigma} = a(\dot{\epsilon}_{vp})\dot{\boldsymbol{\epsilon}}$  where  $\boldsymbol{\sigma}$  is the stress tensor,  $\dot{\epsilon}_{vp}$  is the effective viscoplastic strain  
 54 rate,  $a(t) = t^{m-1}$ ,  $m > 0$  is a power law function, and  $\dot{\boldsymbol{\epsilon}}$  is the strain rate tensor. In our model,  
 55 the inertia effects are ignored. This constitutive equation is similar to the non-Newtonian fluid  
 56 flow problems [7] and helps in relating the stress components to the strain-rate components. The  
 57 associated formulation allows a direct imposition of the plastic flow incompressibility constraint,  
 58 i.e., the divergence free condition of the velocity field, [6], [10].

59 The equilibrium system formulates the balance between the external and the internal forces. The  
 60 finite element discrete analog of the equilibrated equations results in a (nonlinear) system, with  
 61 the velocity and the pressure on the nodal mesh points, which are functions of time, as the un-  
 62 known quantities. Having computed the solution for the system, we can update the configuration  
 63 using an explicit time-stepping scheme. Finite element schemes for flow formulations were pre-  
 64 sented with different time-stepping schemes for thin-sheet-forming processes in previous studies,  
 65 see, e.g., [23], [36].

66 As mentioned above, the rolling deformation process is a complicated problem and cannot be  
 67 treated easily via numerical calculations, even though the whole numerical computation is driven  
 68 to a steady-state. One reason for this is the strong nonlinearities that appear in the model. Apart  
 69 from the nonlinear terms that appear in the constitutive law, nonlinear inequality constraints  
 70 also arise on the contact interface due to the imposition of bilateral contact conditions (non-  
 71 penetration constraint) and due to form of the friction laws. Another reason is related to the  
 72 changing of the contact interface points. The contact interface is not constant and pre-defined,  
 73 but changes with the time evolution of the problem since the plate moves between the rollers.  
 74 Consequently, after every time step, we need to re-determine the contact interface and the points  
 75 that live on it. We note that the points located before the entrance of the roller gap are mov-  
 76 ing with a lower velocity than the rollers, while the points after the roller gap are moving with  
 77 higher velocity. This change of the plate velocity brings about analogous inlet/outlet boundary  
 78 constraints (inequalities). Different techniques exist for incorporating these boundary constraints.  
 79 Here, we employ the penalty term technique [21], [9]. Furthermore, due to the velocity change of  
 80 the plate, the tangential friction stresses,  $\mathbf{T}_\tau$ , change direction. The location of the neutral point,  
 81 where the relative velocity between the tangential plate and the roll are equal, i.e., the point  
 82 where  $\mathbf{T}_\tau$  changes direction, is not a priori known. However, it “moves“ through the initial time  
 83 steps and finally becomes fixed during the steady-state computations.

84 Another important point in rolling problems is the modeling (and discretization) of the interfacial  
 85 friction phenomena between the rollers and the metal plate. Due to the complexity of the rolling  
 86 process, simplified friction laws, such as Coulomb’s law, are not appropriate and thus, more ad-  
 87 vanced friction laws must be used for obtaining accurate results (see, e.g., [5] and the references  
 88 therein). The use of an appropriate friction law is important for simulating the problem because  
 89 the friction forces play a crucial role in driving the metal plate into the roller gap and conse-  
 90 quently influences the behavior of the deformed plate. In this work, viscoplastic type friction laws  
 91 are used which are compatible with the viscoplastic constitutive equation (see the form of the  
 92 friction stress vector  $\mathbf{T}_\tau$  in (2.29a)). These types of laws have been used in many works, showing  
 93 better results compared to other more classical friction laws (see [5], [20]).

94 We apply a continuous finite element method for the discretization of the problem, where the  
 95 interface friction conditions are consistently adapted in the discrete variational form. The bound-  
 96 ary and the non-penetration constraints are enforced by introducing penalty terms. For this, we  
 97 construct a set of nonlinear penalty functions that are added in the equilibrium system and are

98 activated when the inequality constraints are not satisfied [9]. The convergence properties of these  
 99 penalty terms are not investigated in this work.

100 The neutral point is a singular point for the viscoplastic friction law and simple regularization  
 101 techniques are applied by adding a small perturbation constant in the friction law. In order to  
 102 compute the tangential friction stresses, it is necessary to estimate the value of the associated  
 103 friction parameter (denoted by  $a_f$  in (2.29a)). For some simplified problems, laboratory exper-  
 104 iments can usually provide a good estimation of this parameter. In general, its value depends  
 105 on several quantities such as temperature, surface roughness, material properties, etc. Specific  
 106 results on the estimation of the friction parameter for hot rolling are difficult to obtain due to the  
 107 nature of the rolling process [5]. In the current work, an estimation formula of the parameter is  
 108 provided through a proof of giving stability bounds for the problem. In the numerical examples,  
 109 the coefficient  $a_f$  is computed at every time step using this estimate. The new update is used for  
 110 the next time step computation.

111 A numerical investigation of the rolling process is presented by performing an extensive series of  
 112 numerical simulation tests. We investigate the variations of the relative velocity and the values  
 113 of the normal/tangential stresses with respect to the sensitivity strain parameter.

114 The remainder of this paper is outlined as follows: the derivation of the viscoplastic constitutive  
 115 law is discussed in Section 2. Issues related to the variational formulation of a simple deformation  
 116 problem and the contact-friction conditions are also discussed in Section 2, thus preparing all the  
 117 necessary concepts required for modeling the rolling problem, which is given in Section 3. The  
 118 procedure of finite element discretization along with the steps followed for estimating the friction  
 119 parameter are also given in this section. Several numerical tests performed for a numerical study  
 120 of the rolling problem are presented in Section 4. The paper closes with the conclusions given in  
 121 Section 5.

## 122 2 The model problem

### 123 2.1 Preliminaries

We use a standard notation throughout this work. Let  $\Omega$  be a bounded Lipschitz domain in  $\mathbb{R}^d$ ,  $d = \{2, 3\}$ . For a differentiable function  $\phi : \Omega \rightarrow \mathbb{R}$  the gradient at  $x = (x_1, \dots, x_d)$  is the vector  $\nabla\phi(x) = (\frac{\partial\phi(x)}{\partial x_1}, \dots, \frac{\partial\phi(x)}{\partial x_d})$ . We denote by  $L^p(\Omega)$ ,  $p > 1$ , the Lebesgue space of measurable functions  $\phi : \Omega \rightarrow \mathbb{R}$  such that  $\int_{\Omega} |\phi(x)|^p dx < \infty$  endowed with the norm  $\|\phi\|_{L^p(\Omega)} = (\int_{\Omega} |\phi(x)|^p dx)^{\frac{1}{p}}$ . Let  $\alpha = (\alpha_1, \dots, \alpha_d)$  be a multi-index of non-negative integers  $\alpha_1, \dots, \alpha_d$  with degree  $|\alpha| = \sum_{j=1}^d \alpha_j$ . For any  $\alpha$ , we define the differential operator  $D^\alpha = \partial/\partial x_1^{\alpha_1} \dots \partial/\partial x_d^{\alpha_d}$ , and denote the standard Sobolev spaces by

$$W^{\ell,p}(\Omega) = \{\phi \in L^p(\Omega) : D^\alpha\phi \in L^p(\Omega), \text{ for all } |\alpha| \leq \ell\}, \quad (2.1)$$

endowed with the following norms

$$\|\phi\|_{W^{\ell,p}(\Omega)} = \left( \sum_{0 \leq |\alpha| \leq \ell} \|D^\alpha\phi\|_{L^p(\Omega)}^p \right)^{\frac{1}{p}}.$$

where the derivatives in (2.1) are considered in the weak sense. The definition of the spaces in (2.1) is naturally extended to the vector value functions  $\phi = (\phi_1, \dots, \phi_d)$ . For simplicity, we denote the associated spaces by  $\mathbf{W}^{\ell,p}$ . We refer to Ref. [1] for a complete description of the Sobolev spaces. For a later use, we define the vector space

$$\mathbf{V} := \mathbf{W}^{\ell,p}(\Omega), \text{ with } \ell \geq 1, p = m + 1, \quad (2.2)$$

where  $m$  is the exponent parameter in the viscoplastic model, see (2.20).

Let  $p > 1$ , we define its conjugate  $q$  by the relation  $\frac{1}{p} + \frac{1}{q} = 1$ . We recall Hölder's and Young's inequalities

$$\left| \int_{\Omega} \phi_1 \phi_2 dx \right| \leq \|\phi_1\|_{L^p(\Omega)} \|\phi_2\|_{L^q(\Omega)} \quad \text{and} \quad \left| \int_{\Omega} \phi_1 \phi_2 dx \right| \leq \frac{\epsilon}{p} \|\phi_1\|_{L^p(\Omega)}^p + \frac{1}{q\epsilon} \|\phi_2\|_{L^q(\Omega)}^q, \quad (2.3)$$

124 that hold for all  $\phi_1 \in L^p(\Omega)$  and  $\phi_2 \in L^q(\Omega)$  and for any fixed  $\epsilon \in (0, \infty)$ , [1]. We shall use  
 125 the summation convention, according to which repeated indices indicate a summation from 1 up  
 126 to the dimension of the involved vectors. We will frequently use the double contracted product  
 127 between tensors, namely  $A : B = \sum_{i,j} A_{ij} B_{ij}$ , which results in a scalar. The double contracted  
 128 product defines an inner product between tensors.

## 129 2.2 Notations

Let the set  $\Omega \subset \mathbb{R}^d$ ,  $d \in \{2, 3\}$  be the (reference) configuration of a body consisting of a viscoplastic material. We are interested in describing the deformation of the body subjected to body forces  $\mathbf{f}$  and surface tractions,  $\mathbf{T}$ , which are applied to a part of the body surface. As usual the displacement of each point  $\mathbf{x} \in \Omega$  is denoted by  $\mathbf{u} = (u_1, \dots, u_d)$  and the components  $\varepsilon_{ij}$  of the linear (infinitesimal) strain tensor  $\boldsymbol{\varepsilon}$  produced by  $\mathbf{u}$  are given by  $\varepsilon_{ij} = \frac{1}{2}(\frac{\partial u_i}{\partial x_j} + \frac{\partial u_j}{\partial x_i})$ , with  $1 \leq i \neq j \leq d$ . The components of the velocity  $\mathbf{v} = (v_1, \dots, v_d)$  of the material points at a certain time  $t$  are defined by  $v_i = \frac{\partial u_i}{\partial t}$  and the components of the strain rate tensor  $\dot{\boldsymbol{\varepsilon}}$  are given by

$$\dot{\varepsilon}_{ij} = \frac{1}{2} \left( \frac{\partial v_i}{\partial x_j} + \frac{\partial v_j}{\partial x_i} \right), \quad \text{with } 1 \leq i \neq j \leq d. \quad (2.4)$$

Here we suppose, see next sections, that there exist a viscoplastic potential  $\varphi := \varphi(\dot{\boldsymbol{\varepsilon}})$  such that the stress-strain constitutive relations can be derived by

$$\sigma_{ij} = \frac{\varphi(\dot{\boldsymbol{\varepsilon}})}{\partial \dot{\varepsilon}_{ij}}, \quad (2.5)$$

130 where  $\sigma_{ij}$  are the components of the stress tensor  $\boldsymbol{\sigma}$ . In the deformation problems here the  
 131 hydrostatic pressure has very little effect and the strain rates and the associated stresses are  
 132 deviatoric. We introduce the deviatoric stress tensor  $\mathbf{s} := \boldsymbol{\sigma} + P\mathbf{I}$  with  $P = \frac{1}{d} \text{trace}(\boldsymbol{\sigma})$ , which has  
 133 a trace equal to 0 and the same principal directions as  $\boldsymbol{\sigma}$ . Also, we note that we consider small  
 134 strain cases and the linear strain tensor can be split into an elastic part  $\boldsymbol{\varepsilon}_e$  and a plastic part  $\boldsymbol{\varepsilon}_p$ ,  
 135 i.e.,  $\boldsymbol{\varepsilon} = \boldsymbol{\varepsilon}_e + \boldsymbol{\varepsilon}_p$ .

## 136 2.3 Plasticity criterion and stress-strain relations

It is known from the studies of small strain plasticity problems [4], that the materials after being subjected to a stress  $\sigma$  beyond the yield stress  $\sigma_0$ , they have plastic deformation behavior, which means that permanent deformations of the material exist when we return back to the unstressed state. The behavior of the materials before reaching the yield stress point  $\sigma_0$  is elastic without the evolution of plastic strains. In a general situation when a stress tensor  $\boldsymbol{\sigma}$  is defined on a point  $\mathbf{x}$  of the material, we would like to have a criterion in order to verify if the related stresses are below  $\sigma_0$  (elastic region) or above  $\sigma_0$  (plastic region). Mathematically this means that we need to formulate an inequality constraint that will depend on the stresses  $\sigma$  and  $\sigma_0$ . Below we give this inequality and also derive the basic equations in plasticity, i.e., yield condition, flow rule, stress-strain relation, following the von Mises viscoplasticity framework, see e.g., [2], [4], [26]. Let us consider the principal - axial stress case for  $d = 3$  and define

$$J_2 = \frac{1}{2} s_{ij} s_{ij}, \quad (2.6)$$

where  $s_{ij}$  are the components of  $\mathbf{s}$  defined by

$$\mathbf{s} = \begin{pmatrix} \dots & & 0 \\ 0 & \sigma_{ii} - \frac{1}{3}(\sigma_{11} + \sigma_{22} + \sigma_{33}) & \\ 0 & & \dots \end{pmatrix} \quad (2.7)$$

Using the above relations, we introduce the von Misses plasticity yield function, [4],

$$\Phi(\boldsymbol{\sigma}, \sigma_0) = \left(3J_2(\mathbf{s}(\boldsymbol{\sigma}))\right)^{\frac{1}{2}} - \sigma_0 := \sigma_{ef} - \sigma_0, \quad (2.8)$$

137 where we let  $\sigma_{ef} := \sqrt{3}\sqrt{J_2(\mathbf{s}(\boldsymbol{\sigma}))}$  to be the effective stress. Based on (2.8), we define the plasticity  
 138 surface  $\mathcal{E} = \{\boldsymbol{\sigma} : \Phi(\boldsymbol{\sigma}, \sigma_0) = 0\}$ . A stress  $\boldsymbol{\sigma}$  such that  $\Phi(\boldsymbol{\sigma}, \sigma_0) < 0$  lies in the elastic domain  
 139 and for a  $\boldsymbol{\sigma} \in \mathcal{E}$  a plastic deformation can occur. We note that here we focus on the solution of  
 140 purely viscoplastic problems, and hence we neglect the elastic strain component in the additive  
 141 decomposition of the strain rate tensor  $\dot{\boldsymbol{\epsilon}} = \dot{\boldsymbol{\epsilon}}_e + \dot{\boldsymbol{\epsilon}}_p$ , i.e., we set  $\dot{\boldsymbol{\epsilon}}_e = 0$ .

Having defined the plasticity criterion (2.8), we proceed to express the plastic flow rule, i.e., the particular form of the plastic strain rate tensor  $\dot{\boldsymbol{\epsilon}}_p$ . The derivation of the plastic flow rule is based on the principle of the maximum work: given a stress  $\boldsymbol{\sigma}$  which verifies the plasticity criterion,  $\boldsymbol{\sigma} \in \mathcal{E}$ , and its associated strain rate tensor  $\dot{\boldsymbol{\epsilon}}_p$ , then for every other stress  $\hat{\boldsymbol{\sigma}}$  such that  $\Phi(\hat{\boldsymbol{\sigma}}, \sigma_0) \leq 0$ , we have

$$\hat{\boldsymbol{\sigma}} : \dot{\boldsymbol{\epsilon}}_p \leq \boldsymbol{\sigma} : \dot{\boldsymbol{\epsilon}}_p, \quad (2.9)$$

where the double contracted product between tensors has been used. Based on (2.9), we can conclude that  $\dot{\boldsymbol{\epsilon}}_p$  must be proportional to the exterior normal  $\frac{\partial \Phi}{\partial \boldsymbol{\sigma}}$  on the  $\mathcal{E}$

$$\dot{\boldsymbol{\epsilon}}_p = \lambda \frac{\partial \Phi}{\partial \boldsymbol{\sigma}}, \quad \text{or componentwise} \quad \varepsilon_{p,ij} = \lambda \frac{\partial \Phi}{\partial \sigma_{ij}}, \quad (2.10)$$

142 where the scalar parameter  $\lambda$  determines the plastic rate, for more details we refer to [4],[2].

Note that (2.10) specifies the form of  $\dot{\boldsymbol{\epsilon}}_p$  and the direction of the plastic flow through the term  $\frac{\partial \Phi}{\partial \boldsymbol{\sigma}}$ . By (2.7) we have that

$$\frac{\partial s_{ij}}{\partial \sigma_{mn}} = \begin{cases} \frac{d-1}{d} := \frac{2}{3} & \text{if } i = j = m = n \\ \frac{1}{d} := \frac{1}{3} & \text{if } i = j \neq m = n \end{cases}. \quad (2.11)$$

Now recalling (2.8) and that  $\sigma_{ef} = \sqrt{\frac{3}{2}} \|\mathbf{s}\|$  after performing few computations we can show that

$$\dot{\varepsilon}_{p,ij} = \frac{\partial \Phi}{\partial \sigma_{jj}} = \sum_i \frac{\partial \sigma_{ef}}{\partial s_{ii}} \frac{\partial s_{ii}}{\partial \sigma_{jj}} = \sqrt{\frac{3}{2}} \frac{1}{\|\mathbf{s}\|} \frac{1}{3} (2s_{jj} - \sum_{i \neq j} s_{ii}) = \sqrt{\frac{3}{2}} \frac{s_{ij}}{\|\mathbf{s}\|} := \sqrt{\frac{3}{2}} n_i, \quad 1 \leq i, j \leq 3, \quad (2.12)$$

where  $\mathbf{n} = (n_1, n_2, n_3)$  is the outward normal vector on  $\mathcal{E}$ , and its principal directions  $n_i$ ,  $i = 1, 2, 3$  coincide with those of  $\mathbf{s}$ . The flow rule results in

$$\dot{\boldsymbol{\epsilon}}_p = \lambda \sqrt{\frac{3}{2}} \frac{\mathbf{s}}{\|\mathbf{s}\|} = \lambda \sqrt{\frac{3}{2}} \mathbf{n}, \quad (2.13)$$

and by (2.13) we immediately have that  $\lambda = \sqrt{\frac{2}{3}} \|\dot{\boldsymbol{\varepsilon}}_p\| := \dot{\varepsilon}_{ep}$ . On the other hand, we have that the stress  $\mathbf{s}$  satisfies the plasticity criterion (2.8), thus  $\sigma_0^2 = \sigma_{ef}^2 = \frac{3}{2} s_{ij} s_{ij}$ , and then by (2.10) and (2.13) we get that

$$\dot{\varepsilon}_{p,ij} = \dot{\varepsilon}_{ep} \sqrt{\frac{3}{2}} \frac{1}{\sqrt{\frac{2}{3}} \sigma_0} s_{ij} = \frac{3}{2} \left( \frac{\dot{\varepsilon}_{ep}}{\sigma_0} \right) s_{ij}, \quad (2.14a)$$

or in tensor form

$$\dot{\boldsymbol{\varepsilon}}_p = \sqrt{\frac{3}{2}} \left( \frac{\|\dot{\boldsymbol{\varepsilon}}_p\|}{\sigma_0} \right) \mathbf{s}. \quad (2.14b)$$

143 *Remark 2.1.* In many metal plasticity models, the history of the plastic deformation must be  
 144 taken into account. It is usually defined using the effective plastic strain  $\varepsilon_{ep} = \int_0^T \dot{\varepsilon}_{ep}(t) dt$ , where  
 145  $\dot{\varepsilon}_{ep}$  is the effective plastic strain rate, which is defined in the previous analysis. The effective plastic  
 146 strain is often used to characterize the inelastic properties of the material, [2], [4].

147 *Remark 2.2.* An evolution of the plastic strain rate can be accompanied by an evolution of the  
 148 strength of the plastic threshold,  $\sigma_0$ . The increase in the plastic threshold after its initial value is  
 149 called work hardening. The hardening behavior of the material generally depends on the history  
 150 of the plastic deformation. In this case,  $\sigma_0$  is taken to be a function of the effective plastic strain,  
 151 i.e.,  $\sigma_0 := \sigma_0(\varepsilon_{ep})$  and the plasticity criterion (2.8) takes the form  $\Phi(\mathbf{s}, \sigma_0) = \sigma_{ef}^2 - \sigma_0^2(\varepsilon_{ep})$ . The  
 152 analytical form of  $\sigma_0(\varepsilon_{ep})$  is determined by rheological tests, see the discussion in [22] and [4].

153 *Remark 2.3.* As a continuation of Remark 2.2, we note that for the stresses where  $\sigma_{ef}$  is less than  
 154 the flow stress  $\sigma_0(\varepsilon_{ep})$ , the behavior of the physical problem is elastic. When the effective stress  
 155  $\sigma_{ef}$  reaches the value of  $\sigma_0(\varepsilon_{ep})$ , the strain rate partly contains a plastic strain rate.

## 156 2.4 The viscoplastic constitutive law

It has been well verified through experimental observations that in hot metal plastic deformations the stress-strain relation is appropriately described by using power-law rules, i. e., the stress tensor exhibits a power-law dependence on the strain rate tensor (Norton-Hoff viscoplastic constitutive relations), [10], [24]. In several cases in the derivation and in the further analysis of the models of metal forming problems, the introduction and definition of the plasticity yield surface  $\mathcal{E}$ , which separates the elastic from the plastic domain, see (2.8), is not necessary. For example, in many metal forging procedures at high temperatures the values of  $\sigma_0$  are very small, and the metals behave as flowing under weak stresses. Consequently we can consider  $\sigma_0$  to be zero or can neglect  $\sigma_0$  from the formulation of the model. Thus, Norton-Hoff viscoplastic models with  $\sigma_0 = 0$  are widely used, [6], [24], [10]. In their uniaxial form the flow rule is given by

$$\sigma = K |\dot{\varepsilon}_p|^m, \quad (2.15)$$

where  $K > 0$  is a temperature dependent material parameter and the coefficient  $m > 0$  is the strain rate sensitivity coefficient.

We extend the notions of the previous paragraph and denote by  $\dot{\boldsymbol{\varepsilon}}_{vp}$  the viscoplastic strain rate tensor and define the effective viscoplastic strain by

$$\dot{\varepsilon}_{vp} = \sqrt{\frac{2}{3} \dot{\varepsilon}_{vp,ij} \dot{\varepsilon}_{vp,ij}} := \sqrt{\frac{2}{3}} \|\dot{\boldsymbol{\varepsilon}}_{vp}\|, \quad (2.16)$$

where the summation convention for the repeated indices has been used. Now the von-Mises based multi-axial generalization of (2.15) has the form, [4]

$$\sigma_{ef} = K 3^{\frac{m+1}{2}} \dot{\epsilon}_{vp}^m, \quad (2.17)$$

and recalling (2.14) we have that

$$\dot{\epsilon}_{vp,ij} = \frac{3}{2} \frac{\dot{\epsilon}_{vp}}{\sigma_{ef}} s_{ij}, \quad \text{or in tensor form} \quad \dot{\boldsymbol{\epsilon}}_{vp} = \frac{3}{2} \frac{\dot{\epsilon}_{vp}}{\sigma_{ef}} \mathbf{s}. \quad (2.18)$$

Inserting (2.17) into (2.18) we get

$$\dot{\boldsymbol{\epsilon}}_{vp} = \frac{3}{2} \frac{\dot{\epsilon}_{vp}}{K 3^{\frac{m+1}{2}} \dot{\epsilon}_{vp}^m} = \frac{\sqrt{3}^{1-m}}{2K} \dot{\epsilon}_{vp}^{1-m} \mathbf{s}, \quad (2.19)$$

where we can finally obtain a multidimensional generalization of the Norton-Hoff type viscoplastic constitutive relation given in (2.15)

$$\mathbf{s} = 2K (\sqrt{3} \dot{\epsilon}_{vp})^{m-1} \dot{\boldsymbol{\epsilon}}_{vp}. \quad (2.20)$$

157 It should be noted that, by comparing the models of interest given in (2.14b) and (2.20), it can be  
 158 seen that the stress multiplier consists of a non-linear function of the effective viscoplastic strain  
 159 rate  $\dot{\epsilon}_{vp}$ , which follows a power-law rule, and essentially relates the stress variations to the strain  
 160 rate. This multiplier corresponds to a viscosity term, [7], which for  $m = 1$  is equal to  $2K$  (linear  
 161 relation). In the numerical computations, the regularized form

$$\mathbf{s} = 2K \sqrt{3}^{m-1} (\epsilon_0^2 + \dot{\epsilon}_{vp}^2)^{\frac{m-1}{2}} \dot{\boldsymbol{\epsilon}}_{vp} \quad (2.21)$$

162 is used for avoiding deviations with very small numbers.

163 *Remark 2.4.* We can see by (2.21) that the deviatoric stress tensor  $\mathbf{s}$  can be derived by the  
 164 viscoplastic potential  $\varphi := \frac{K}{m+1} (\sqrt{3} \dot{\epsilon}_{vp})^{m+1}$ , i.e.,  $\mathbf{s} = \frac{\partial \varphi}{\partial \dot{\boldsymbol{\epsilon}}_{vp}}$ , see also (2.5).

## 165 2.5 Equilibrium equations

We recall the notations given in Section 2.2, and let  $\Omega$  denote the bounded domain in the space occupied by a viscoplastic continuum. The boundary  $\partial\Omega$  consists of two parts,  $\Gamma_D$  and  $\Gamma_N$ , i.e.,  $\partial\Omega = \Gamma_D \cup \Gamma_N$ , with  $|\Gamma_D| > 0$ . Let  $\Omega$  be acted upon by an interior force  $\mathbf{f} = (f_1, f_2, f_3)$  and a boundary force  $\mathbf{T} = (T_1, T_2, T_3)$  act on  $\Gamma_N$ . On the boundary  $\Gamma_D$ , the displacement  $\mathbf{u}$  is fixed, e.g.,  $\mathbf{u} = 0$ , and consequently the point velocity is  $\mathbf{v} = 0$ . The true internal stresses inside the body are described by the symmetric stress tensor  $\boldsymbol{\sigma}$ . As mentioned in the previous sections we focus on the problem of incompressible materials (note again that  $\mathbf{s} = \boldsymbol{\sigma} + P\mathbf{I}$ , with  $P := -\frac{1}{d} \text{trace}(\boldsymbol{\sigma})$ , is deviatoric and  $\text{trace}(\mathbf{s}) = \text{div } \mathbf{v} = 0$ ). We recall the vector space  $\mathcal{V} = \mathbf{W}^{\ell,p}(\Omega)$ , with  $\ell \geq 1, p = m + 1$ , given in (2.2) and further define

$$\mathcal{V}_{0,D} := \{\boldsymbol{\phi} \in \mathcal{V} : \text{div } \boldsymbol{\phi} = 0, \boldsymbol{\phi} = 0 \text{ on } \Gamma_D\}. \quad (2.22)$$

Using these notations, we can write the equilibrium of  $\Omega$  at a time  $t$  as,

$$\int_{\Omega} \mathbf{s}(\mathbf{v}) : \dot{\boldsymbol{\epsilon}}(\boldsymbol{\phi}) \, dx - \int_{\Omega} P\mathbf{I} : \dot{\boldsymbol{\epsilon}}(\boldsymbol{\phi}) \, dx - \int_{\Gamma_N} \mathbf{T} \cdot \mathbf{v} \, dS = \int_{\Omega} \mathbf{f} \cdot \boldsymbol{\phi} \, dx, \quad \forall \boldsymbol{\phi} \in \mathcal{V}_{0,D}. \quad (2.23)$$

We neglect the volume forces, e.g., the inertia and the gravity forces, and set  $\mathbf{f} = 0$ . Now, we consider the hydrostatic pressure  $P$  as the Lagrange multiplier for enforcing the constraint

$\operatorname{div}(\mathbf{v}) = 0$ , and set the following saddle point problem: find the velocity  $\mathbf{v} \in \mathbf{V}_D := \{\boldsymbol{\phi} \in \mathbf{V} : \boldsymbol{\phi} = 0 \text{ on } \Gamma_D\}$  and the pressure  $P \in L^q(\Omega)$  with  $q = \frac{p}{p-1}$  such that

$$\begin{cases} \int_{\Omega} \mathbf{s}(\mathbf{v}) : \dot{\boldsymbol{\epsilon}}(\boldsymbol{\phi}) \, dx - \int_{\Omega} P \mathbf{I} : \dot{\boldsymbol{\epsilon}}(\boldsymbol{\phi}) \, dx - \int_{\Gamma_N} \mathbf{T} \cdot \boldsymbol{\phi} \, dS & = 0, \\ \int_{\Omega} \operatorname{div}(\mathbf{v}) Q \, dx & = 0, \end{cases} \quad (2.24)$$

166 for all  $\boldsymbol{\phi} \in \mathbf{V}$  and  $Q \in L^q(\Omega)$ . We emphasize that the unknown  $\mathbf{v}$  appears implicitly in (2.24) by  
 167 means of the tensor  $\mathbf{s}$ , which is a function of the strain rate tensor  $\dot{\boldsymbol{\epsilon}}$ , which in turn is a function  
 168 of  $\mathbf{v}$ . The pressure  $P$  is a Lagrange multiplier for the incompressibility condition  $\operatorname{div}(\mathbf{v}) = 0$ .

## 169 2.6 Contact between two bodies

Following the above formulation, we consider the friction contact problem between two viscoplastic bodies, which can undergo a finite deformation. The two bodies in their initial (reference) configuration are given by  $\Omega_1$  and  $\Omega_2$  with  $\Omega_i \subset \mathbb{R}^2$ ,  $i = 1, 2$ . The associated boundaries  $\partial\Omega_i$  are divided into three disjoint parts: (i)  $\Gamma_D^{(i)}$ , where the displacements are prescribed, (ii)  $\Gamma_N^{(i)}$ , where the stresses are prescribed, and (iii) the common contact part  $\Gamma_C$ , where contact conditions will be defined. We assume  $|\Gamma_D^{(i)}| > 0$  for  $i = 1, 2$ . On  $\Gamma_C$ , we define the normal vector  $\mathbf{n}_{12}$  in the direction towards the interior of  $\Omega_2$ . We use the subindex  $i = 1$  or  $i = 2$  to denote the restriction on the associated domains  $\Omega_1$  or  $\Omega_2$  respectively. The stress vectors on  $\Gamma_C$  are given by

$$\mathbf{T}_1 = \boldsymbol{\sigma}_1 \cdot \mathbf{n}_{12}, \quad \mathbf{T}_2 = \boldsymbol{\sigma}_2 \cdot (-\mathbf{n}_{12}). \quad (2.25)$$

170 Both stresses act on the contact area  $\Gamma_C$  and are opposite by obeying the principle of action and  
 171 reaction, i.e.,  $\mathbf{T}_1 = \boldsymbol{\sigma}_1 \cdot \mathbf{n}_{12} = -\boldsymbol{\sigma}_2 \cdot \mathbf{n}_{21} = -\mathbf{T}_2$ . Each of the stress vectors  $\mathbf{T}_i$ ,  $i = 1, 2$  can be  
 172 decomposed with respect to  $\mathbf{n}_{12}$  into a normal  $\mathbf{T}_{i,n}$  and a tangential component  $\mathbf{T}_{i,\tau}$ , [32], for  
 173 example for  $i = 1$

$$\mathbf{T}_{1,n} = (\mathbf{T}_1 \cdot \mathbf{n}_{12})\mathbf{n}_{12}, \quad \mathbf{T}_{1,\tau} = \mathbf{T}_1 - (\mathbf{T}_1 \cdot \mathbf{n}_{12})\mathbf{n}_{12}. \quad (2.26)$$

The scalar  $\sigma_n := \sigma_n^1 = \mathbf{T}_1 \cdot \mathbf{n}_{12} < 0$  (in compression) is the normal stress, and the tangential vector  $\mathbf{T}_{i,\tau}$  (orthogonal to  $\mathbf{n}_{12}$ ) is associated with the friction forces on  $\Gamma_C$ , [32]. On the interface  $\Gamma_C$ , we define the velocity difference  $\Delta\mathbf{v} = \mathbf{v}_1 - \mathbf{v}_2$  and the relative slip velocity

$$\mathbf{v}_s = \Delta\mathbf{v} - (\Delta\mathbf{v} \cdot \mathbf{n}_{12})\mathbf{n}_{12}. \quad (2.27)$$

Along the contact interface  $\Gamma_C$  the bodies can not interpenetrate, i.e.,  $\Omega_1 \cap \Omega_2 = \emptyset$ . The continued bilateral contact between  $\Omega_1$  and  $\Omega_2$  without penetration is usually expressed by  $\Delta\mathbf{v} \cdot \mathbf{n}_{12} = 0$ . In view of this, during the numerical computations, we have to take into account the conditions

$$g_n := \Delta\mathbf{v} \cdot \mathbf{n}_{12} = 0, \quad (2.28a)$$

$$\sigma_n < 0, \quad (2.28b)$$

174 across the contact area  $\Gamma_C$ . Several methods have been presented in the literature for incorporating  
 175 the contact constraints (2.28) in the variational form and in the finite element discrete analogue,  
 176 see, e.g., [21], [32], and the references therein. Here, we will apply the penalty method, where an  
 177 extra term is introduced to penalize the velocity of having a normal penetration component.

178 *Remark 2.5.* Note that for the two normal stresses  $\sigma_n^1 = \mathbf{T}_1 \cdot \mathbf{n}_{12} = -(\boldsymbol{\sigma}_2 \cdot \mathbf{n}_{21}) \cdot \mathbf{n}_{12} = (\boldsymbol{\sigma}_2 \cdot \mathbf{n}_{21}) \cdot$   
 179  $\mathbf{n}_{21} = \sigma_n^2$  holds. By this condition we can also infer that  $\sigma_n^1 = \sigma_n^2$ , which indicates that the normal  
 180 interface stresses are compressive, and so this requires  $\sigma_n^1 < 0$ , compare with (2.28b).

181 *Remark 2.6.* The sticking (frictionless) contact conditions are  $\Delta\mathbf{v} = 0$  and  $(\boldsymbol{\sigma}_2 \cdot \mathbf{n}_{21}) + (\boldsymbol{\sigma}_1 \cdot \mathbf{n}_{12}) = 0$ .

**Frictional contact** In general, for deriving a relation which can describe in an complete way the interfacial friction phenomena and the friction tangential stresses  $\mathbf{T}_\tau$ , apart from the normal stress and the relative tangential velocity, one must also take into account some other physical parameters, such as temperature, surface roughness, interaction of chemical processes, etc, which are often described in micro-scales. It is clear that the mixing and the interrelation of all these quantities will lead to a complicated model. In simple practical applications the classical Coulomb's law, which includes the normal stress and the relative velocity, or Tresca's law, which includes a constant shear strength, are usually used, [20], [5], [32], [2]. For the purposes of this work, we introduce a nonlinear friction law that is compatible with (2.20) (without threshold), which describes the tangential effects and can be considered as a generalization of Tresca's law. Accordingly we let

$$\mathbf{T}_\tau = -a_f K |\mathbf{v}_s|^{m-1} \mathbf{v}_s, \quad (2.29a)$$

182 where  $a_f$  is a friction parameter, which is estimated below, and  $K$  is a material parameter which  
 183 can be dependent on stain hardening, see (2.15). In this case the interface conditions are completed  
 184 by adding the bilateral contact condition (2.28a).

### 185 3 Application to strip rolling

186 As we mentioned in the previous sections, the hot strip rolling process is very important and  
 187 most commonly used technique in engineering metal-forming, since it is used for the design of  
 188 rotating machine parts, gears, ball bearings, etc, [10], [8], [15], [18], [33]. In the flat rolling process  
 189 the thickness of a flat metal plate (strip) is reduced by passing it between two counter-rotating  
 190 cylinders (rollers) which have a fixed distance, as shown in an illustration in Fig. 1(a). For an  
 191 analysis of rolling problems we refer to [16], [15], [33]. We use the relations and the forms that  
 192 were derived in the previous section for describing the rolling process. In our study the metal  
 193 plate is the viscoplastic material that occupies  $\Omega_1$  and the rollers are the rigid tools  $\Omega_2$ . We take  
 194 advantage of the  $X$ -axis symmetry of the problem and consider the upper half of the problem,  
 195 see Fig. 1(a). We focus on deriving the problem formulation for the case of having maximum  
 196 length of the contact interface  $\Gamma_C$ , and setting symmetry boundary conditions along the  $X$ -axis.  
 197 A schematic illustration of the boundary parts of  $\partial\Omega_1$  with the associated boundary conditions  
 198 is given in Fig. 1(b). Since all the quantities below are related to the deformation of the plate,  
 199 i.e.,  $\Omega_1$ , we remove the corresponding index from their notation.

#### 200 3.1 The boundary value problem for the rolling contact

During the rolling process, the metal plate is moving between the two rollers and the tangential friction forces that exist across the interface  $\Gamma_C$  drive the plate in to the gap. At the inlet points, i.e., points on the left vertical boundary  $\Gamma_{in}$ , see Fig. 1(b), the velocity of the plate  $\mathbf{v}_{in}$  is lower than the tangential velocity of the rollers. As the plate moves in to the gap of the rollers, it is compressed and this increases its velocity due to the conservation of mass. At the neutral point,  $P_{neutral}$  on  $\Gamma_C$ , the relative velocity is  $\mathbf{v}_s = 0$  and finally the plate exits the roller gap with velocity  $\mathbf{v}_{out}$ , which is greater than the velocity of the rollers, see Fig. 1(b). Since additional elastic phenomena are ignored, these remarks lead to the following constraints on the boundary

$$\mathbf{v} \cdot (-\mathbf{n}_{in}) \leq |\mathbf{U}_{roll,\tau}| \text{ on } \Gamma_{in} \quad \text{and} \quad \mathbf{v} \cdot (\mathbf{n}_{out}) \geq |\mathbf{U}_{roll,\tau}| \text{ on } \Gamma_{out}, \quad (3.1)$$

where  $|\mathbf{U}|_{roll,\tau}$  is the measure of the tangential velocity of the roller. We apply penalty method techniques, [11], [9], for treating the boundary constraints given in (3.1), the bilateral contact constraint in (2.28a) and the constraint for preventing the vertical motion on the free-stress

boundary parts denoted by  $\Gamma_0$  in Fig. 1(b). For a function  $f : \bar{\Omega}_1 \cup \bar{\Omega}_2 \rightarrow \mathbb{R}$  we define  $[f(x)]_+ := \max\{f(x), 0\}$  and introduce the penalty functionals

$$\Psi_{in}(\mathbf{v}) = \frac{\gamma_{in}}{2} \int_{\Gamma_{in} \cup \Gamma_{0L}} ([\mathbf{v} \cdot (-\mathbf{n}_{in}) - |\mathbf{U}_{roll,\tau}|]_+)^2 ds, \quad (3.2a)$$

$$\Psi_{out}(\mathbf{v}) = \frac{\gamma_{out}}{2} \int_{\Gamma_{out} \cup \Gamma_{0R}} ([-\mathbf{v} \cdot (\mathbf{n}_{out}) + |\mathbf{U}_{roll,\tau}|]_+)^2 ds, \quad (3.2b)$$

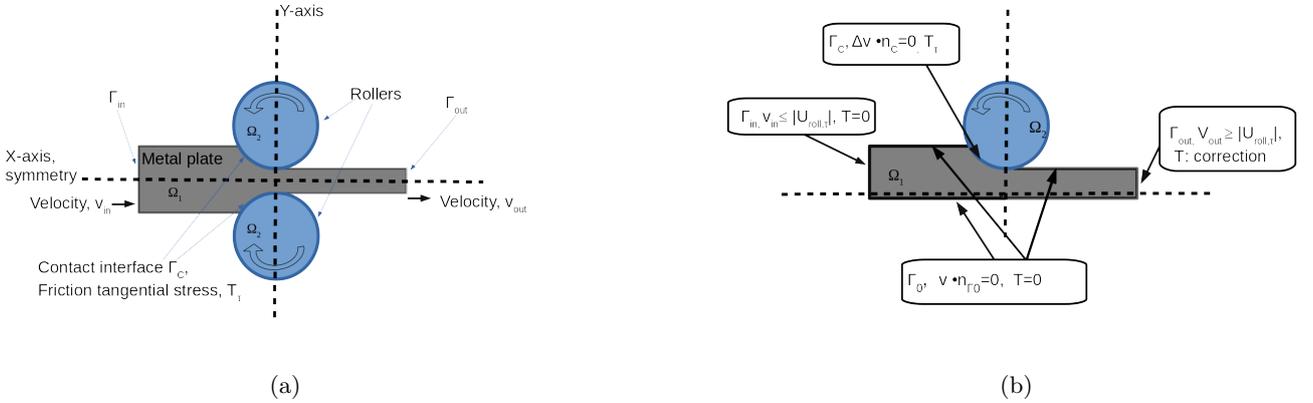
$$\Psi_{v_C}(\mathbf{v}) = \frac{\gamma_{v_C}}{2} \int_{\Gamma_C} (|\Delta \mathbf{v} \cdot \mathbf{n}_C|^2 - Tol)_+^2 ds, \quad (3.2c)$$

$$\Psi_{\Gamma_0}(\mathbf{v}) = \frac{\gamma_{v_0}}{2} \int_{\Gamma_0} (\mathbf{v} \cdot \mathbf{n}_{\Gamma_0})^2 ds, \quad (3.2d)$$

where  $\gamma_{in}$ ,  $\gamma_{out}$ ,  $\gamma_{v_C}$  and  $\gamma_{v_0}$  are the penalty parameters (their values will be specified later),  $\mathbf{n}_{in}$ ,  $\mathbf{n}_{out}$ ,  $\mathbf{n}_C$  and  $\mathbf{n}_{\Gamma_0}$  are the normals on the associated boundary parts, and the tolerance  $Tol \approx 1.E - 05$ . In (3.2) the boundary parts  $\Gamma_{0L}$  and  $\Gamma_{0R}$  are the free-stress boundary parts with points  $(x, y)$  such that  $\Gamma_{0L} := \{(x, y) \in \Gamma_0 : x \leq x_{neutral\ point}\}$  and  $\Gamma_{0R} := \{(x, y) \in \Gamma_0 : x \geq x_{neutral\ point}\}$ . Thus, utilizing (2.24), (2.29a) and (3.2) we can express the following penalty formulation for the rolling problem: find  $\mathbf{v}$  and  $P$  such that

$$\begin{cases} \int_{\Omega_1} \mathbf{s}(\mathbf{v}) : \dot{\boldsymbol{\varepsilon}}(\boldsymbol{\phi}) dx - \int_{\Omega_1} P \mathbf{I} : \dot{\boldsymbol{\varepsilon}}(\boldsymbol{\phi}) dx - \int_{\Gamma_0 \cup \Gamma_{in}} \mathbf{T}_0 \cdot \boldsymbol{\phi} dS - \int_{\Gamma_C} \mathbf{T}_\tau \cdot \boldsymbol{\phi} dS + \int_{\Gamma_{out}} \mathbf{T}^* \cdot \boldsymbol{\phi} dS \\ - \gamma_{in} \int_{\Gamma_{in} \cup \Gamma_{0L}} [\mathbf{v} \cdot (-\mathbf{n}_{in}) - |\mathbf{U}_{roll,\tau}|]_+ \boldsymbol{\phi} \cdot \mathbf{n}_{in} dS - \gamma_{out} \int_{\Gamma_{out} \cup \Gamma_{0R}} [-\mathbf{v} \cdot (\mathbf{n}_{out}) + |\mathbf{U}_{roll,\tau}|]_+ \boldsymbol{\phi} \cdot \mathbf{n}_{out} dS \\ - \gamma_{v_0} \int_{\Gamma_0} (\mathbf{v} \cdot \mathbf{n}_{\Gamma_0}) \boldsymbol{\phi} \cdot \mathbf{n}_{\Gamma_0} dS - \gamma_{v_C} \int_{\Gamma_C} [|\Delta \mathbf{v} \cdot \mathbf{n}_C|^2 - Tol]_+ \boldsymbol{\phi} \cdot \mathbf{n}_C dS = 0, \\ \int_{\Omega} \text{div}(\mathbf{v}) Q dx = 0, \end{cases} \quad (3.3)$$

201 for all test functions  $\boldsymbol{\phi} \in \mathbf{V}$  and  $Q \in L^q(\Omega)$  where  $\Gamma_0 \cup \Gamma_{in}$  are the free stress boundary parts,  
 202 i.e.,  $\mathbf{T}_0 = 0$ , and  $\mathbf{T}^*$  on  $\Gamma_{out}$  describes the free stress condition plus a penalty correction term that  
 203 prevents the points  $\mathbf{x} \in \Gamma_{out}$  to move vertically (along  $Y$ -axis) to the main movement of the plate  
 along  $X$ -axis.



**Fig. 1.** Schematic diagram of the rolling problem, (a) the rollers with the deformed plate and the  $X$ -axis symmetry line, (b) the boundary parts on  $\partial\Omega_1$  with the associated boundary conditions.

### 205 3.2 Finite element discretization

We use finite element methodology to discretize (3.3). Each domain  $\Omega_i$ ,  $i = 1, 2$  is subdivided into a collection  $T_h^i$  of (conforming) triangular mesh elements  $\{K\}$  such that  $\Omega_i = \cup_{K \in T_h^i} K$ . We set  $T_h = T_h^1 \cup T_h^2$ . We define the mesh size  $h_i$ ,  $i = 1, 2$  to be the length of the maximum edge of  $K \in T_h^i$ . On each  $T_h^i$  we consider the space  $V_{h,\kappa}^i$  of continuous piece-wise polynomials of order  $\kappa$ , i.e.,

$$V_{h,\kappa}^i = \{v_h : v_h \in C^0(\Omega_i), v_h|_K \in \mathbb{P}^\kappa(K), K \in T_h^i\}, \quad (3.4)$$

where  $C^0$  is the space of continuous functions and  $\mathbb{P}^\kappa$  is the space of polynomials of order  $\kappa$ . Let  $\{\phi_j^i\}_{j=1}^{N_i}$ ,  $i = 1, 2$  be a corresponding basis of each  $V_{h,\kappa}^i$ , which is defined through the  $N_i$  nodes of  $T_h^i$ . Then, for every  $v_h \in V_{h,\kappa}^i$  we have

$$v_h = \sum_{j=1}^{N_i} a_j \phi_j^i(x, y), \quad (3.5)$$

where  $a_i$  are the related  $N_i$  nodal values of  $v_h$ , see details in [14], [3]. In view of (3.4) the discrete analogue of (3.3) is: find  $\mathbf{v}_h := (v_{1,h}, v_{2,h}) \in V_{h,\kappa}^1 \times V_{h,\kappa}^1$  and  $P_h \in V_{h,l}^1$  such that

$$\begin{cases} \int_{\Omega_1} \mathbf{s}(\mathbf{v}_h) : \dot{\boldsymbol{\varepsilon}}(\boldsymbol{\phi}_h) dx - \int_{\Omega_1} P_h \mathbf{I} : \dot{\boldsymbol{\varepsilon}}(\boldsymbol{\phi}_h) dx - \int_{\Gamma_C} \mathbf{T}_{\tau,h} \cdot \boldsymbol{\phi}_h dS + \int_{\Gamma_{out}} \mathbf{T}_{0,h}^* \cdot \boldsymbol{\phi}_h dS \\ - \gamma_{in} \int_{\Gamma_{in} \cup \Gamma_{0L}} [\mathbf{v}_h \cdot (-\mathbf{n}_{in}) - |\mathbf{U}_{roll,\tau}|]_+ \boldsymbol{\phi}_h \cdot \mathbf{n}_{in} dS \\ - \gamma_{out} \int_{\Gamma_{out} \cup \Gamma_{0R}} [-\mathbf{v}_h \cdot (\mathbf{n}_{out}) + |\mathbf{U}_{roll,\tau}|]_+ \boldsymbol{\phi}_h \cdot \mathbf{n}_{out} dS \\ - \gamma_{v_0} \int_{\Gamma_0} (\mathbf{v}_h \cdot \mathbf{n}_{\Gamma_0}) \boldsymbol{\phi}_h \cdot \mathbf{n}_{\Gamma_0} dS - \gamma_{v_C} \int_{\Gamma_C} [|\Delta \mathbf{v}_h \cdot \mathbf{n}_C|^2 - Tol]_+ \boldsymbol{\phi}_h \cdot \mathbf{n}_C dS \\ \int_{\Omega} \operatorname{div}(\mathbf{v}_h) Q_h dx \end{cases} = 0, \quad (3.6)$$

206 for  $\boldsymbol{\phi}_h := (\phi_{1,h}, \phi_{2,h}) \in V_{h,\kappa}^1 \times V_{h,\kappa}^1$  and  $Q_h \in V_{h,l}^1$ .

207 Note that different meshing of the contact  $\Gamma_C$  can occur through the nodes on  $\partial\Omega_i$ . We can have  
 208 matching mesh nodes and non-matching mesh nodes. For producing the finite element solutions,  
 209 we need to first identify the nodes that lay on the contact interface, and the rest nodes which are  
 210 located away from the contact area. In the most realistic cases with finite deformations, we prefer  
 211 to perform the computations without requiring matching mesh restrictions across  $\Gamma_C$ . We note  
 212 here that the tangential velocities of the two bodies are not equal, and the domain  $\Omega_2$  (roller)  
 213 has a fixed known tangential velocity. Due to the frictional shear stresses (2.29a), the variation  
 214 of the relative interface velocity  $\mathbf{v}_s$  and the conditions in (3.1), the mesh nodes enter and leave  
 215  $\Gamma_C$  with different rates. The non-penetration of the bodies on  $\gamma_C$  is enforced in the numerical  
 216 computations through the penalty terms related to (2.28a). In (3.6) the spaces  $V_{h,\kappa}^1$  and  $V_{h,l}^1$  has  
 217 been chosen to satisfy the in-sup condition, i.e., in our numerical examples we set  $l = k - 1$ , [28].  
 218 The configuration is updated following an explicit procedure, (see Section 3.3 below).

### 219 3.3 Explicit time integration

As the material passes through the gap of the rollers the rolling becomes a steady state process. We find numerically the steady-state solution by using an explicit time-stepping algorithm for updating the configuration. The whole time period of the study of the problem, let say  $T = [0, T_F]$ , is partitioned into small time increments  $[t_n, t_{n+1}]$ ,  $t_{n+1} = t_n + \Delta t$  with fixed time step  $\Delta t$ . In a

sequential repeating procedure, we solve (3.6) at time  $t_n$  and then using the computed velocity  $\mathbf{v}_{h,t_n}$  we find (approximately) the new configuration  $\Omega_{t_{n+1}}$  at  $t_{n+1}$  by updating the coordinates  $\mathbf{x}^{node}$  of the mesh nodes

$$\mathbf{x}_{t_{n+1}}^{node} = \mathbf{x}_{t_n}^{node} + \Delta t \mathbf{v}_{h,t_n}. \quad (3.7)$$

In (3.7)  $\mathbf{x}_{t_{n+1}}^{node}$  are the new coordinates of the mesh nodes of  $\Omega_{t_{n+1}}$  which is going to be used as the new reference configuration. The computations are being repeated for the next configurations following the procedure described previously. We solve (3.6) on  $\Omega_{t_{n+1}}$  and applying (3.7) compute the coordinates of the mesh nodes of  $\Omega_{t_{n+2}}$ .

### 3.4 Implementation remarks

The final nonlinear algebraic system resulting from (3.6) is solved by using a simple Picard iterative method of the type

$$\text{for every time step } t_n, n = 1, 2, \dots \text{ solve} \quad (3.8a)$$

$$\mathbf{J}(\mathbf{U}_n^i) \mathbf{U}_n^{i+1} = \mathbf{b}(\mathbf{U}_n^i, \mathbf{U}_{roll}), \quad \text{for } i = 0, 1, \dots, i_{max}, \quad (3.8b)$$

$$\text{when } \|\mathbf{U}_n^i - \mathbf{U}_n^{i+1}\| \leq 1.E - 05 \quad \text{set } \mathbf{U}_n^{i+1} := \mathbf{U}_n^{solution}, \quad (3.8c)$$

$$\text{and move to the next time step,} \quad (3.8d)$$

where for every interior iteration  $i$  the vector  $\mathbf{U}_n^i$  includes the unknown degrees of freedom of  $\mathbf{v}_h$  and  $P_h$ ,  $\mathbf{J}$  is the linearized matrix which includes the non-linearities of (2.21) and (2.29a) computed using the previous iteration  $\mathbf{U}_n^i$ ,  $\mathbf{b}$  is the right hand side and  $\mathbf{U}_n^{solution}$  is the final solution for the time step  $t_n$ . In the numerical tests we have set  $i_{max} = 20$ , but as we approach the steady-state the criterion in (3.8c) is satisfied in less than 10 iterations.

The value for the small constant  $\varepsilon_0$  in (2.21) is set to 1.E-04. In order to avoid a singular behavior of the stress form (2.21), we solve the associated linear system in (3.6), by setting  $m = 1$ , during the first iteration  $i = 0$  at the first time step  $t_1$ , and then we use the corresponding solutions to compute the entries of  $\mathbf{J}(\mathbf{U}_1^{i=1})$  in (3.8b). The solution of the linear system provides an initial guess for the velocity which is corrected during the next iterative steps of the scheme given in (3.8).

Within the application of the finite element scheme (3.6) at every time increment, it is necessary to know which nodes are in contact and which are on free stress boundary parts at every time step. Having known the mesh nodes which are on  $\Gamma_C$  at  $t_n$  we need to find the points which are in contact at the end of the time increment, i.e., after computing the new configuration using (3.7). In fact this is mainly related to the position of the upper boundary points which enter or leave from the contact zone. We suppose that the points which are or which are not on  $\Gamma_C$  remain in the same state throughout the interior steps of the iterative procedure (3.8). The control for changing the boundary characterization of the upper part of the boundary points is performed after the node update obtained from (3.7). Due to the small variations of  $\Delta \mathbf{v}_s \cdot \mathbf{n}_C$  at the entrance/exit contact points, we prefer to apply the geometric condition

$$\text{if } \|\mathbf{x}_{t_{n+1}}^{node} - \Pi(\mathbf{x}_{t_{n+1}}^{node})\| < 1.E - 05, \text{ then } \mathbf{x}_{t_{n+1}}^{node} \in \Gamma_C, \quad (3.9a)$$

$$\text{if } \|\mathbf{x}_{t_{n+1}}^{node} - \Pi(\mathbf{x}_{t_{n+1}}^{node})\| \geq 1.E - 05, \text{ then } \mathbf{x}_{t_{n+1}}^{node} \text{ is a free stress point,} \quad (3.9b)$$

where  $\Pi(\mathbf{x}_{t_{n+1}}^{node})$  is the orthogonal projection of  $\mathbf{x}_{t_{n+1}}^{node}$  to the roll surface.

The interface  $\Gamma_C$  is approximated by linear elements and through this approximation large overlap and/or gap regions can exist between the mesh points of the plate and the roller surface (specially for coarse mesh tests). In order to avoid the existence of these large overlap/gap regions, every ten time steps we apply a post-processing correction for the locations of the internal mesh

241 nodes of  $\Gamma_C$ , i. e., we correct the position of the nodes based on maximum distance function given  
 242 in (3.9a).

243 Due to the change in the boundary conditions on the last contact point  $M = (1, 0.2)$  the  
 244 normal pressure can exhibit an oscillatory behavior. A post-processing correction is applied on  
 245 the elements after the last exit contact point in order to eliminate the oscillations of the normal  
 246 pressure.

247 The maximum distance function given in (3.9a) helps in defining the time step  $\Delta t$ . In our  
 248 computations we have chosen  $\Delta t = \frac{h_C}{6}$ , where  $h_C$  is the mesh size across  $\Gamma_C$ .

249 As the plate is driven out of the roller gap, the penalty functionals given in (3.2) are applied for  
 250 all the corresponding boundary parts of  $\partial\Omega_1$  which are formed after the rolling, see Fig. 1(b) and  
 251 Fig. 2(b). Since the neutral point  $P_{neutral}$  has no fixed location until reaches to the steady-state,  
 252 in our computations we specify  $\Gamma_{0L}$  and  $\Gamma_{0R}$  with respect to the last contact point  $M = (1, 0.2)$ ,  
 253 i. e.,  $\Gamma_{0L} := \{(x, y) \in \Gamma_0 : x \leq 1\}$  and  $\Gamma_{0R} := \{(x, y) \in \Gamma_0 : x \geq 1\}$ .

The values of the parameters in (3.2) are given in the following Table 1

	$I = [0.75, 0.775]$			
parameter	$x \in I$	$x \notin I$	$\Gamma_{in}$	$\Gamma_{out}$
$\gamma_{v_0}$	$\frac{10}{h}$	$\frac{25}{h}$	-	-
$\gamma_{v_C}$	$\frac{10}{h}$	$\frac{25}{h}$	-	-
$\gamma_{v_{in}}$	-	-	$\frac{2}{h^{m+1}}$	-
$\gamma_{v_{out}}$	-	-	-	$\frac{2}{h^{m+1}}$

**Table 1.** The values of the penalty parameters.

254

### 255 3.5 Stability bounds and an estimation of $a_f$

The value of the friction parameter  $a_f$  in (2.29a) is in general unknown. Below we try to give an estimate for  $a_f$  by providing stability bounds for the scheme (3.6). Consider (3.6) without the introduction of the boundary constraints and free stress  $\mathbf{T} = 0$ . We set  $m - 1 = p - 2$ , with  $p > 1$ . Setting  $\phi_h = \mathbf{v}_h$  and  $Q_h = P_h$  in variational formulation (3.6), and recalling that  $\mathbf{s} = 2K(\sqrt{3}\dot{\epsilon}_{eq})^{m-1}\dot{\epsilon}$  and  $\mathbf{T}_\tau = -a_f K |\mathbf{v}_s|^{m-1} \mathbf{v}_s$ , we can obtain

$$\int_{\Omega_1} 2K(\sqrt{3}\dot{\epsilon}_{eq})^{p-2} \dot{\epsilon} : \dot{\epsilon} dx + \int_{\Gamma_C} a_f |\mathbf{v}_{h,s}|^{p-2} \mathbf{v}_{h,s} \cdot \mathbf{v}_h ds = 0. \quad (3.10)$$

Using the decomposition  $\mathbf{v}_h = \mathbf{v}_{h,\tau} + \mathbf{v}_{h,n}$  with  $\mathbf{v}_{h,\tau} \cdot \mathbf{v}_{h,n} = 0$  in (3.10) we get

$$\begin{aligned} \int_{\Omega_1} C_K(\dot{\epsilon}_{eq})^p dx &= - \int_{\Gamma_C} a_f |\mathbf{v}_{h,s}|^{p-2} \mathbf{v}_{h,s} \cdot (\mathbf{v}_{h,\tau} + \mathbf{v}_{h,n} - \mathbf{U}_{roll,\tau} + \mathbf{U}_{roll,\tau}) ds \\ &= - \int_{\Gamma_C} a_f |\mathbf{v}_{h,s}|^{p-2} \mathbf{v}_{h,s} \cdot (\mathbf{v}_{h,s} + \mathbf{v}_{h,n} + \mathbf{U}_{roll,\tau}) ds \\ &= - \left( \int_{\Gamma_C} a_f |\mathbf{v}_{h,s}|^p ds + \int_{\Gamma_C} a_f |\mathbf{v}_{h,s}|^{p-2} \mathbf{v}_{h,s} \cdot (\mathbf{v}_{h,n} + \mathbf{U}_{roll,\tau}) ds \right) \\ &= - \int_{\Gamma_C} a_f |\mathbf{v}_{h,s}|^p ds + \int_{\Gamma_C^-} a_f |\mathbf{v}_{h,s}|^{p-1} |\mathbf{U}_{roll,\tau}| ds - \int_{\Gamma_C^+} a_f |\mathbf{v}_{h,s}|^{p-1} |\mathbf{U}_{roll,\tau}| ds, \\ &= - \int_{\Gamma_C} a_f |\mathbf{v}_{h,s}|^p ds + \int_{\Gamma_C} a_f |\mathbf{v}_{h,s}|^{p-1} |\mathbf{U}_{roll,\tau}| ds - 2 \int_{\Gamma_C^+} a_f |\mathbf{v}_{h,s}|^{p-1} |\mathbf{U}_{roll,\tau}| ds, \end{aligned} \quad (3.11)$$

where we used that  $\mathbf{v}_{h,s}$  and  $\mathbf{U}_{roll,\tau}$  are parallel, and defined  $\Gamma_C^+ := \{(x, y) \in \Gamma_C : \mathbf{v}_{h,s} \cdot \mathbf{U}_{roll,\tau} = |\mathbf{v}_{h,s}| |\mathbf{U}_{roll,\tau}|\}$  and analogously  $\Gamma_C^- := \{(x, y) \in \Gamma_C : \mathbf{v}_{h,s} \cdot \mathbf{U}_{roll,\tau} = -|\mathbf{v}_{h,s}| |\mathbf{U}_{roll,\tau}|\}$ . Note that

since  $\mathbf{v}_{h,s}$  and  $\mathbf{U}_{roll,\tau}$  are parallel there exists a  $\lambda : \lambda(x)$ ,  $|\lambda| < 1$  such that  $\mathbf{v}_{h,s} = \lambda \mathbf{U}_{roll,\tau}$ . Furthermore note that on  $\Gamma_C^-$  it holds that  $|\mathbf{v}_{h,\tau}| < |\mathbf{U}_{roll,\tau}|$ . On the other hand the difference  $\mathbf{v}_{h,\tau} - \mathbf{U}_{roll,\tau}$  across  $\Gamma_C^+$  is very small, and since  $p - 1 > 0$ , we then may suppose

$$|\mathbf{v}_{h,s}|^{p-1} \approx 0 \quad \text{on} \quad \Gamma_C^+. \quad (3.12)$$

Now, in (3.11) we apply inequalities (2.3) with  $p - 1 = \frac{p}{q}$ , and find

$$\begin{aligned} \int_{\Omega_1} C_K(\dot{\varepsilon}_{eq})^p dx &\geq \int_{\Gamma_C} a_f (|\lambda|^{p-1} - |\lambda|^p) |\mathbf{U}_{roll,\tau}|^p ds \\ &\quad - c_{1,\varepsilon} \int_{\Gamma_C^+} |\mathbf{v}_{h,s}|^p ds - c_{2,\varepsilon} \int_{\Gamma_C^+} a_f^p |\mathbf{U}_{roll,\tau}|^p ds. \end{aligned} \quad (3.13)$$

Finally, we select the parameter  $c_{2,\varepsilon}$  in (3.13) sufficiently small, e. g.,  $c_{2,\varepsilon} = \frac{1}{2a^{p-1}} \min \{|\lambda|^{p-1} - |\lambda|^p\}$  and obtain

$$\int_{\Omega_1} C_K(\dot{\varepsilon}_{eq})^p dx + c_{1,\varepsilon} \int_{\Gamma_C^+} |\mathbf{v}_{h,s}|^p ds \geq \int_{\Gamma_C} \frac{a_f}{2} \min \{|\lambda|^{p-1} - |\lambda|^p\} |\mathbf{U}_{roll,\tau}|^p ds. \quad (3.14)$$

256 The lower bound given in (3.14) seems not so convenient for estimating the parameter  $a_f$ . In view  
257 of (3.12) we can consider that the corresponding integrals on  $\Gamma_C^+$  have negligible contribution to  
258 the right hand side in (3.11). Thus based on that we omit these integrals through our computations  
259 and using further the inequalities [3],  $\sum_K \|\nabla \mathbf{v}_h\|_{L^p(K)} \leq \frac{C}{h} \sum_K \|\mathbf{v}_h\|_{L^p(K)}$  for  $\mathbf{v}_h \in V_{h,\kappa}^1 \times V_{h,\kappa}^1$ , and  
260  $(|a|^2 + |b|^2)^{\frac{p}{2}} \leq 2^{p-1}(|a|^p + |b|^p)$ , we deduce that

$$\frac{C_5}{h} \sum_K \int_K |\mathbf{v}_h|^p dx \geq \int_{\Gamma_C^-} a_f \min \{|\lambda|^{p-1} - |\lambda|^p\} |\mathbf{U}_{roll,\tau}|^p ds \quad (3.15)$$

261 Owing that  $|\mathbf{U}_{roll,\tau}|$  is fixed, the magnitude of  $\min \{|\lambda|^{p-1} - |\lambda|^p\}$  can be estimated from the  
262 values of  $|\mathbf{v}_{h,\tau}|$  and  $|\mathbf{U}_{roll,\tau}|$  on the first touching mesh element in the gap entry. Note again that  
263 we accept that  $\mathbf{v}_{h,\tau}$  and  $\mathbf{U}_{roll,\tau}$  are parallel across the touching element.

264 *Remark 3.1.* The relations (3.13) and (3.14) hold also for the continuous solution  $\mathbf{v}$  of (3.3).

265 *Remark 3.2.* The relation (3.15) provides an estimation of  $a_f$  by means of the discrete kinematic  
266 energy and the measure  $|\mathbf{U}_{roll,\tau}|^p$ .

267 *Remark 3.3.* In the numerical examples we estimate  $a_f$  by (3.15), where the integrals  $\int_K |\mathbf{v}_h|^p dx$   
268 are computed numerically using the values  $\mathbf{U}_n^i$  of the previous iteration, see (3.8), and setting  
269  $C_5 = 2^p$ .

*Remark 3.4.* Applying similar computations as above and using (2.3) we can obtain the upper  
bound

$$\int_{\Omega_1} C_K(\dot{\varepsilon}_{eq})^p dx + c_{1,\varepsilon} \int_{\Gamma_C} a_f |\mathbf{v}_{h,s}|^p \leq c_{2,\varepsilon} \int_{\Gamma_C} a_f |\mathbf{U}_{roll,\tau}|^p ds. \quad (3.16)$$

## 270 4 Numerical examples

271 In this section, we perform numerical examples for the steady state rolling problem. Note that  
272 due to the symmetry of the problem with respect to  $X$ -axis, we consider only the upper half of  
273 the problem. The radius of the roll is  $R = 0.6m$  and the center is the point  $C_{roll} := (X_{roll}, Y_{roll}) =$   
274  $(1, 0.8)$ . The initial length of the metal plate is  $L = 4m$  ( $-3 \leq x \leq 1$ ) and the maximum

275 height  $H = 0.25m$  ( $0 \leq y \leq 0.25$ ), respectively. The contact interface consists of the points  
 276  $\Gamma_C = \{(x, y) : 0.760208428 \leq x \leq 1, 0.2 \leq y \leq 0.25, \text{ and } ((x - X_{Roll})^2 + (y - Y_{roll})^2)^{\frac{1}{2}} = 0.6\}$ .  
 277 Therefore, the thickness of the plate at the entrance is  $0.25m$  and at the exit is  $0.2m$ . An illustration  
 278 of the initial configuration of the problem with the touching zone and uniform mesh for the plate  
 279 are shown in Fig. 2(a). In order to examine the convergence of the numerical solution and the  
 280 mesh-independency, the problem is solved using four different uniform meshes. The corresponding  
 281 mesh sizes with the associated abbreviations are listed in Table 2. The meshes are separated into  
 282 three groups:  $C1$  and  $C2$  are the coarse meshes,  $M1$  and  $M2$  are the middle meshes, while  $F1$   
 283 and  $F2$  are the fine meshes. The mesh size refers to the mesh size of the boundary elements, say  
 284  $h$ , that are in contact with the interface of the initial configuration. For every mesh-test case, we  
 285 start the computation having a maximum length of contact, Fig. 2(a), and use a fixed time step  
 286  $\Delta t = \frac{h}{6}$ .

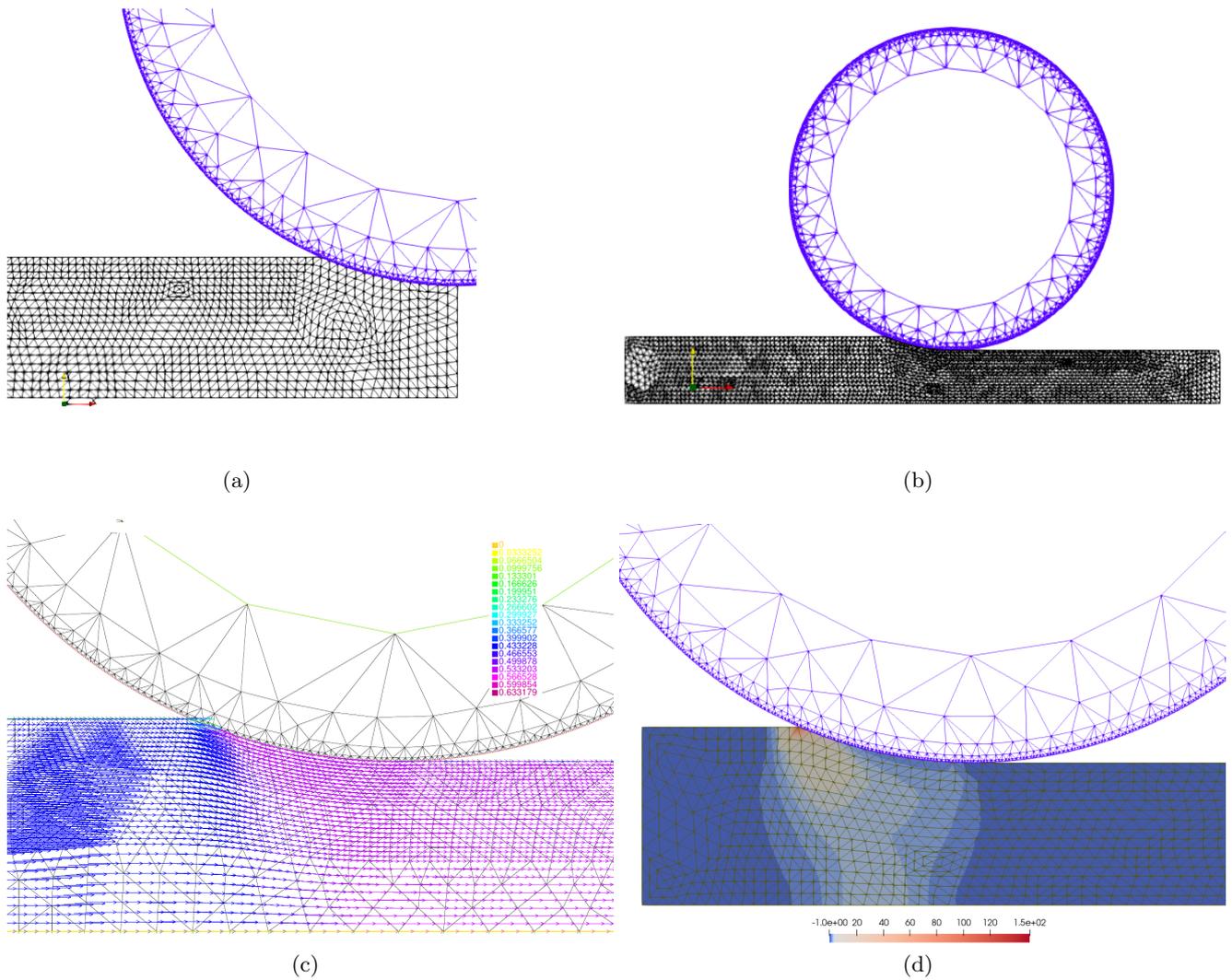
287 During the first initial computations we set  $m = 0.2$  and the angular velocity for the roller, i. e.,  
 288 the rate of change of angle with respect to time, is  $\theta = \frac{\pi}{3}$  per second. In the last examples we solve  
 289 the problem using  $m \in \{0.15, 0.25\}$  and  $\theta = \frac{\pi}{6}$ , and compare the associated numerical results.  
 290 The evolution of the deformation is depicted in Fig. 2. The results have been computed using the  
 291  $C2$  mesh, see Table 2. Fig. 2(b) shows the deformed (after the rolling) and the undeformed (before  
 292 the rolling) parts of the plate at the half computational time. In Fig. 2(c) the velocity vectors  
 293 with the associated mesh in the contact region are plotted, from which, the sharp variation of  
 294 the velocity near the first touching mesh element of the roller entry can be seen. Fig. 2(d) shows  
 295 the deformed plate with the distribution of the magnitude of the stress-rate  $\sigma_{ef} = K3^{\frac{m+1}{2}} \dot{\epsilon}_{vp}^m$ , see  
 296 (2.16). From the figure we can see that the highest values of  $\sigma_{ef}$  are in the area of the entrance  
 297 touching point  $P_{touch} = (0.76, 0.25)$ , where the velocities have sharp variations.

298 The plastic deformation starts near the contact area and becomes greater at the following points.  
 299 It is maximum at the exit point ( $x = 1, y = 0.2$ ), where the permanent deformation is created.  
 300 The variation of the viscoplastic strain rate measure  $\|\dot{\epsilon}(t_n)\|_{L^p}^p := \int_{\Omega_{t_n}} 2K(3\dot{\epsilon}_{vp})^p$ ,  $p = m+1$ , with  
 301 respect to time for  $m \in \{0.15, 0.2, 0.25\}$  is shown in Fig. 3(a). In Fig. 3(b) we present the time  
 302 variation of the relative velocity measure  $\|\mathbf{v}_s(t_n)\|_{L^p}^p := \int_{\Gamma_c} |\mathbf{v}_s(t_n)|^p ds$ . It should be noted that  
 303 for all  $m$  test cases shown in Fig. 3 the same  $\Delta t$  step size has been used. From the two graphs,  
 304 we can see that the variations for all test cases show a periodic behavior with a stable amplitude.  
 305 We can thus consider that the computation has reached the steady-state. As can be seen from  
 306 the figure, the variation of  $\mathbf{v}_s$  for the case of  $m = 0.2$  and  $m = 0.25$  has similar behavior, with its  
 307 maximum value occurring on closed time steps.

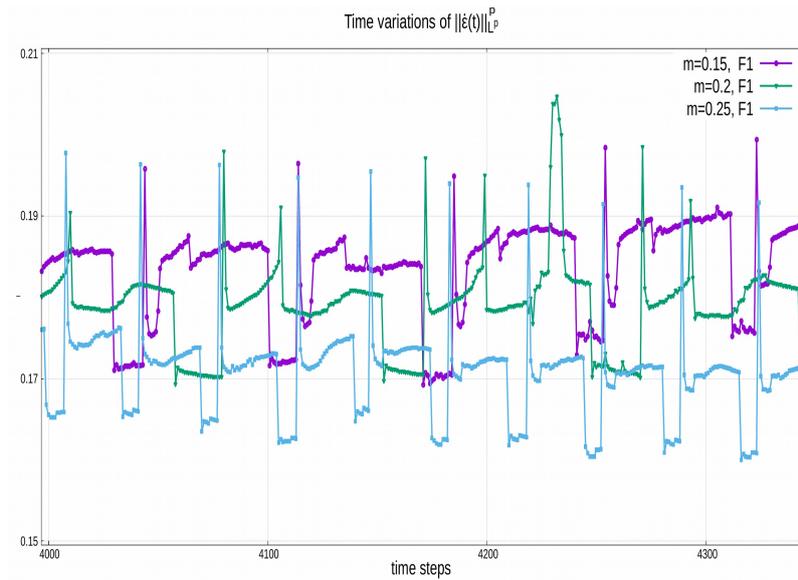
	Meshes					
Name	C1	C2	M1	M2	F1	F2
Mesh size $h$	5.02E-03	4.2E-03	3.7E-03	3.01E-03	2.7E-03	2.02E-03

**Table 2.** The test meshes with the corresponding mesh sizes.

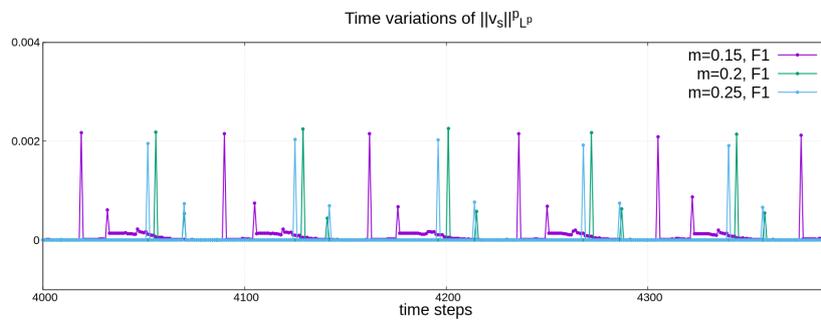
308 *Example 1*,  $m = 0.2$ ,  $\theta = \frac{\pi}{3}$ . In the first example, we start by presenting a short numerical investi-  
 309 gation related to the convergence of the numerical results while refining the mesh. We have solved  
 310 the problem using the meshes given in Table 2 successively. In Fig. 4(a) we plot the variation of  
 311 the velocity component  $v_2$  on the upper boundary points including the interface points. As we  
 312 mentioned above, the velocity  $v_2$  is almost zero at the points before the touching entry point  $P_{touch}$   
 313 of the roller. In a small area after  $P_{touch}$  the velocity  $v_2$  increases sharply (in absolute value) and  
 314 then decreases progressively (decreases almost linearly) reaching zero at the last interface point  
 315 at the exit. As is expected, the coarse meshes cannot capture the sharp gradient of  $v_2$  efficiently  
 316 whereas the middle and fine mesh solutions appear to capture them. The solutions of middle and  
 317 fine meshes have the same behavior without any remarkable differences between them, as can be



**Fig. 2.** (a) The initial configuration with a coarse mesh and the touching line, (b) the deformed configuration and the associated mesh at half computational time, (c) focus on the contact area and the velocity contours, (d) the final deformed plate and the distribution of the stress-rate measure.



(a)



(b)

**Fig. 3.** Example 1: (a) The variation of the strain rate measure with respect to time, (b) the variation of the relative velocity measure with respect to time.

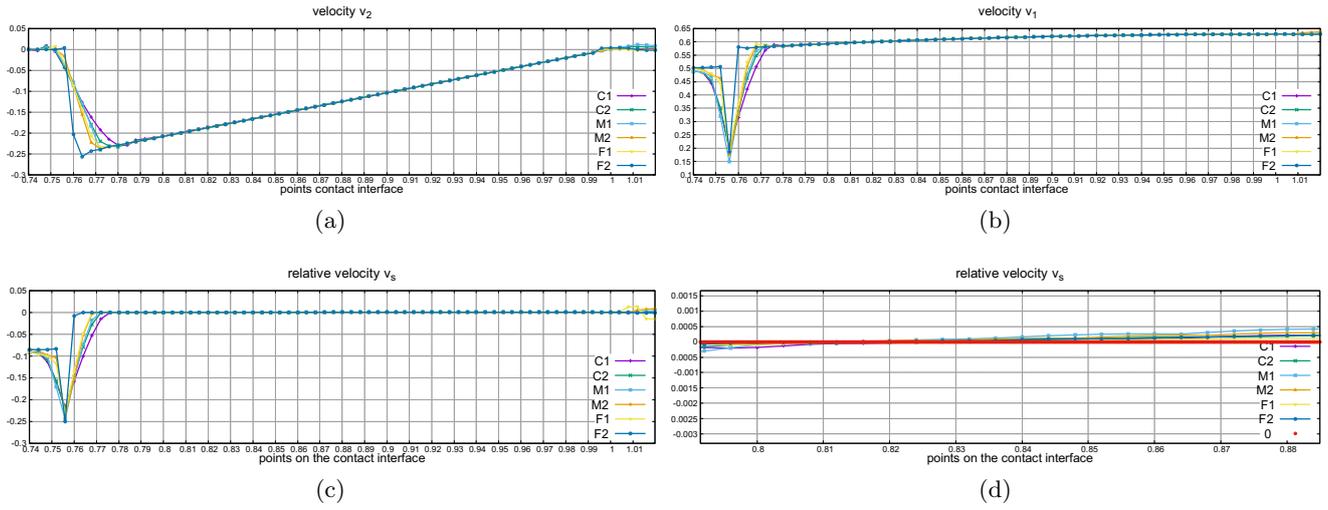
318 seen from the plots close to  $P_{touch}$ .

319 The variation of  $v_1$  for the upper boundary has been plotted in Fig. 4(b) from which a good  
 320 agreement of the numerical solutions is observed. The mesh points before entering into the con-  
 321 tact region have a fixed velocity  $v_1$ , which is reduced in a small zone close to  $P_{touch}$ , compare to  
 322 Fig. 4(a). After passing through this zone, the points accelerate and leave the contact region with  
 323 higher velocity.

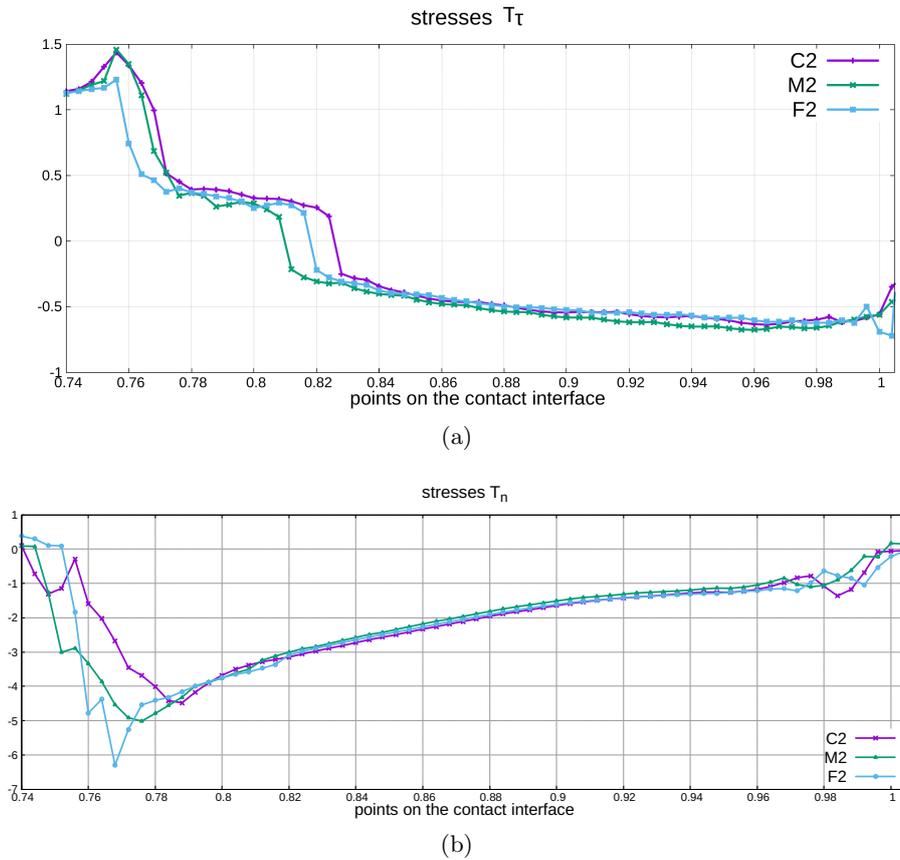
324 The variation of  $\mathbf{v}_s$  at the upper boundary points for the different meshes are shown in Fig 4(c).  
 325 Note that for points outside of  $\Gamma_C$  the computation of  $\mathbf{v}_s$  is not appropriate. For the initial points  
 326 at the contact interface the velocity of the plate points is smaller than the velocity of the roller  
 327 interface points, and thus, we have the negative values of  $\mathbf{v}_s$ . As the plate moves forward its veloc-  
 328 ity increases and finally becomes higher than the velocity of the roller. In any mesh test case the  
 329 relative velocity at the exit points is not so high. We can observe that all the numerical solutions  
 330 have the same behavior in general. The numerical solution corresponding to the  $F2$  mesh captures  
 331 the variations of  $\mathbf{v}_s$  in the neighborhood of  $P_{touch}$  in a better manner. At the neutral point the  
 332 relative velocity between the roller and the plate is zero. The exact location of this point is not a  
 333 priori known, but it takes its final position at the steady-state. From the numerical computations,  
 334 the location of the neutral point has been estimated close to the point  $P_{neutral} \approx (0.818, 0.22754)$ .  
 335 Fig. 4(d) shows the plots of the relative velocity solutions in a region close to the neutral point.  
 336 We can observe that all numerical solutions of  $\mathbf{v}_s$  are zero in the area of the neutral point.  
 337 From a comparison of the graphs above we can say that the numerical solutions computed using  
 338 the middle and the fine meshes can efficiently capture the variations of the velocity, they have  
 339 similar behavior, which is not essentially improved when we move to fine meshes. Thus we can say  
 340 that the middle and fine mesh solutions can provide quite accurate results. In the figures given  
 341 below, as representative examples, we show only three numerical solutions computed using the  
 342 associated  $C2$ ,  $M2$  and  $F2$  meshes.

343 Fig. 5(a) shows the plots of the tangential stress  $\mathbf{T}_\tau \cdot \mathbf{n}_\tau$  over the contact area points. For  
 344 plotting reasons we divide the results with the number 1.3. From the figure, it can be seen that  
 345 small numerical oscillations appear close to the first touching point and close to the exit point  
 346  $(0.2,1)$ , due to the steep changes of the velocity and the boundary conditions. We can see that  
 347 the point where the numerical solution  $C2$  crosses the  $X$ -axis is located at  $x=0.83$ . It is located  
 348 a little further on the right than the point where the two other numerical solutions cross the  
 349  $X$ -axis. The fine mesh related solutions cross the  $X$ -axis close to the neutral point. The normal  
 350 stress variations  $\mathbf{T}_n \cdot \mathbf{n}$  over the contact points are plotted in Fig. 5(b). The solution related to  
 351 the  $F2$  mesh appears to capture the variations at the entrance points more sharply than the two  
 352 other solutions. For the next contact points, the behavior of the solutions is similar and for the  
 353 points after the  $P_{neutral}$  the normal stress variations are similar to the variations of  $\mathbf{T}_\tau \cdot \mathbf{n}_\tau$ . Again,  
 354 we can see small spurious numerical oscillations at the entrance and exit points. It should be  
 355 noted here that during the numerical time step computations, the exact locations of the first and  
 356 last contact points are not fixed and are not a-priori known. These points may be located on the  
 357 boundaries of the edges or may lie in the interior. So during the computations, it is possible to  
 358 have mesh elements with one node on the contact line and the other on the free stress boundary  
 359 parts. This fact is a small reason for the generation of the spurious oscillations.

360 *Example 2*,  $m = 0.2$ ,  $\theta = \frac{\pi}{6}$ . The rolling plate deformation is affected by the radial velocity of  
 361 the roller, [16], [27]. To examine this we have solved the problem by setting  $m = 0.2$ ,  $\theta = \frac{\pi}{6}$  and  
 362 compared the produced solutions with the solutions of the previous examples. In Fig. 6(a) the  
 363 profiles of the relative velocities  $\mathbf{v}_s$  are plotted along the contact points for the mesh  $F1$ . As is  
 364 expected the relative velocity corresponding to  $\theta = \frac{\pi}{3}$  is lower, almost double (in negative values)  
 365 than the velocity of the  $\theta = \frac{\pi}{6}$  case. In Fig. 6(b) the profiles of the relative velocities are plotted  
 366 for the points close to neutral point  $P_{neutral}$ . As can be seen from the figure, the neutral points  
 367 have the same locations for both test cases. The velocity of  $\frac{\pi}{3}$  is lower for the points on the left of

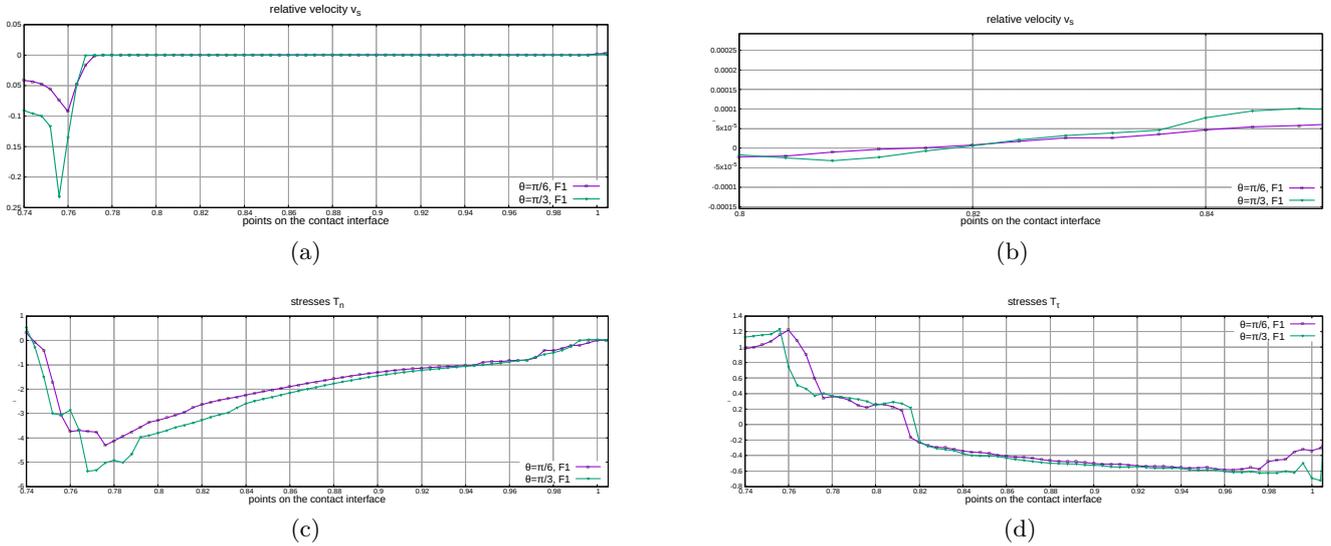


**Fig. 4.** Example 1: (a) The variations of the  $v_2$  velocity component for all meshes, (b) The velocity  $v_1$  for all meshes, (c) The relative velocity  $\mathbf{v}_s$  for all meshes, (d) The relative velocity  $\mathbf{v}_s$  around the neutral point  $P_{neutral} = (0.818, 0.22754)$ , ( $\mathbf{v}_s = 0$ ).



**Fig. 5.** Example 1: (a) The tangential stress  $\mathbf{T}_\tau \cdot \mathbf{n}_\tau$ , (b) The normal contact stress  $\mathbf{T}_n \cdot \mathbf{n}$ .

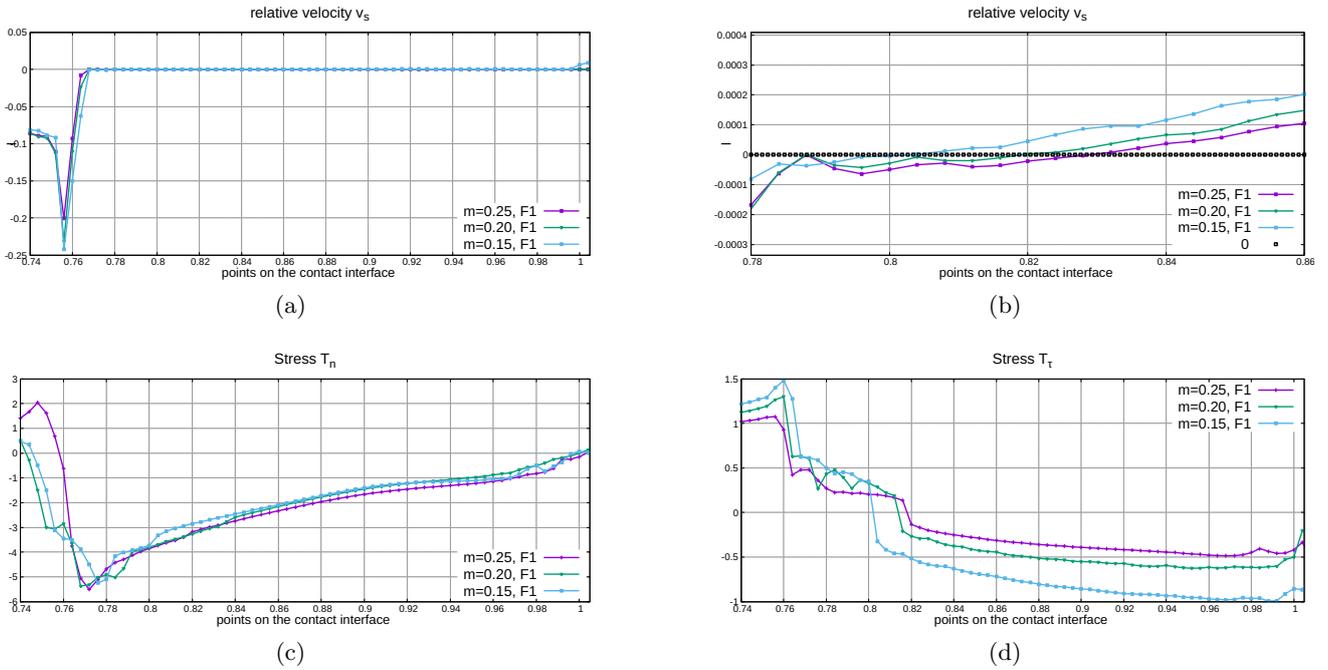
368  $P_{neutral}$ , and increases after  $P_{neutral}$  by progressively double the values as compared to the velocity  
 369 of the  $\frac{\pi}{6}$  case. The variations of the normal stress  $\mathbf{T}_n \cdot \mathbf{n}$  and the tangential stress  $\mathbf{T}_\tau \cdot \mathbf{n}_\tau$  along  
 370 the contact interface are shown in Fig. 6(c) and Fig. 6(d) respectively. As we expected the level  
 371 of the stresses related to the  $\theta = \frac{\pi}{3}$  are higher (in absolute values) than the stresses related to  
 372 the  $\theta = \frac{\pi}{6}$  case. It must be noted here that the numerical solutions of  $\theta = \frac{\pi}{6}$  show less spurious  
 oscillations at the first and last points on the contact line.



**Fig. 6.** Example 2: (a) The relative velocities on the whole contact points for  $\theta = \frac{\pi}{3}$  and  $\theta = \frac{\pi}{6}$ , (b) the variations of the relative velocities in the neighboring of the neutral point, (c) the normal stress  $\mathbf{T}_n \cdot \mathbf{n}$ , and (d) the tangential stress  $\mathbf{T}_\tau \cdot \mathbf{n}_\tau$ .

373

374 *Example 3*,  $m \in \{0.15, 0.2, 0.25\}$ ,  $\theta = \frac{\pi}{3}$ . Rolling is directly affected by the material parameters.  
 375 In order to investigate this effect we have performed computations with different values for the  
 376 sensitivity coefficient of the strain rate by setting  $m \in \{0.15, 0.25\}$ . The computations have been  
 377 performed by keeping the same speed for the roller  $\theta = \frac{\pi}{3}$ . The profiles of the corresponding  
 378 relative velocities for all the cases are given in Fig.7(a). For the points located close to the first  
 379 touching point, the relative velocity decreases (in absolute values) when we increase the exponent  
 380  $m$ , see for example the  $m = 0.25$  test case. For the points slightly to the right of the touching  
 381 point,  $\mathbf{v}_s$  increases by almost the same rate for all  $m$ -test cases. In Fig. 7(b) the variations of the  
 382 relative velocities close to  $P_{neutral}$  are plotted. It can be seen that  $\mathbf{v}_s$  of the test case  $m = 0.15$   
 383 grows faster than the other two cases and crosses the  $X$ -axis at a point which is located slightly  
 384 more to the left of  $P_{neutral}$ , (note here that  $P_{neutral}$  is the neutral point which corresponds to the  
 385  $m = 0.2$  test case). For the rest of the boundary points it remains greater than the other two  
 386 velocities. The relative velocity related to the  $m = 0.25$  case crosses the  $X$ -axis at a point very  
 387 close to  $P_{neutral}$  and has a similar behavior to the relative velocity of the  $m = 0.2$  test case.  
 388 Fig. 7(c) shows the normal stresses over the contact points. For the first points the normal stress  
 389 values of the  $m = 0.15$  case are higher (in absolute value) than the other test cases. For the  
 390 rest of the points the stress curves have similar behavior with only small differences. The plot  
 391 line corresponding to the  $m = 0.25$  case has less oscillations as compared to the other two lines.  
 392 Finally, the curves of the tangential stresses are given in Fig. 7(d). The values related to the  
 393  $m = 0.15$  case are higher (in absolute values) as compared to the other two cases. The position of  
 394 the neutral point has moved slightly on the left compared to the curves for the other two cases,  
 395 compare also with Fig. 7(b). As expected from Figs. 7(a) and (b), the behavior of the tangential  
 396 stresses for the  $m = 0.2$  and  $m = 0.25$  cases is observed to be similar.



**Fig. 7.** Example 3: (a) the relative velocities for the three  $m$ -test cases, (b) the relative velocities close to neutral point, (c) the normal stresses  $\mathbf{T}_n \cdot \mathbf{n}$ , (d) the tangential stress  $\mathbf{T}_\tau \cdot \mathbf{n}_\tau$ .

## 397 Conclusions

398 In this work, viscoplastic mathematical models for the hot rolling process of metal-forming are  
 399 described in detail under the assumption that elastic effects are negligible. After deriving the  
 400 basic constitutive law, the bilateral contact conditions and a viscoplastic type friction law are  
 401 discussed. The associated system of the equilibrium equations has been presented giving special  
 402 emphasis on the construction of the penalty terms that can incorporate the boundary constraints  
 403 of the inlet/outlet velocity as well the bilateral contact conditions in the system. A standard  
 404 finite element scheme with continuous Taylor-Hood polynomial spaces, together with an explicit  
 405 time approach for updating the configuration, have been applied for discretizing the equilibrium  
 406 system. Innovation in this work was the estimation of the friction parameter,  $a_f$ , through the proof  
 407 of stability bounds for the finite element scheme. Moreover, several important aspects, which arise  
 408 during the implementation of the algorithm for solving the rolling problems, are discussed. The  
 409 appropriateness of the whole approach has been investigated by performing several numerical  
 410 tests that concern the estimation of the position of the neutral point, the magnitude of the  
 411 normal/tangential stresses and the variations of the relative tangential velocity. The numerical  
 412 results show that the choice of the strain rate sensitivity coefficient and the roller speed affects  
 413 the magnitude of the normal and tangential stresses and the variations of the relative velocity,  
 414 but not essentially the position of the neutral point.

415 In many realistic contact problems with viscoplastic materials, more detailed modeling of the  
 416 interface friction phenomena and the dependency of the friction coefficient,  $a_f$ , on other variables  
 417 such as temperature and surface roughness is required. Moreover, more advanced finite element  
 418 methodologies and efficient implementation of the algorithm, for example, curved meshes in a  
 419 parallel environment for fast performance computing, is necessary for producing more accurate  
 420 numerical simulations.

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